CONVERSION OF LYCOMING ENGINE FROM 65 HP. 0-145-B2 into 75 HP. 0-145-C2 Serial No. 5780-2

MOONEY M-18-L, N119C Serial No. 51

> REPORT NO. 52 January 23, 1998

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ENGINE:

The LYCOMING Model 0-145-B and -C engines are identical in design with only such changes in construction as to permit different horsepower ratings. The B series is rated at 65 horsepower at 2550 R.P.M. and the C series is rated at 75 horsepower at 3100 R.P.M.

The only difference in the engines is the replacement of the single valve springs and seats in the 65 hp B series engines with the dual valve springs and seats in the 75 hp C series engines.

To convert the B engine to a C engine would entail the removal of the following parts:

45440 Seat–Valve Spring–Lower (Except C Series)45441 Seat–Valve Spring–Upper (Except C Series)45453 Spring–Valve (B Engines Only)

The following parts would then be installed:

- 45471 Seat–Valve Spring–Upper (Dual Springs) (C Engine Only)
- 45472 Seat–Valve Spring–Lower (Dual Springs) (C Engine Only)
- 45473 Spring–Valve–Outer (Dual Springs) (C Engine Only)
- 45474 Spring–Valve–Inner (Dual Springs) (C Engine Only)

PROPELLERS:

There are several approved propellers for the Lycoming powered Mooney M18-L.

Mooney N119C is now equipped with the Flottorp 63L60 propeller, which is a good climb/cruise propeller. It is possible to exceed redline of 2550 R.P.M. in level flight to 8,500 ft. altitude. The primary purpose of this proposed modification is to raise the redline, thereby improving the performance of the aircraft, both in level flight and descending from altitude.

Sensenich has three approved propellers in their line, a cruise and two climb propellers. It is possible that one of the climb propellers, operating at a higher R.P.M., would further improve the performance, both climb and cruise.

HANDBOOK OF INSTRUCTIONS WITH PARTS CATALOG

LYCOMING

MODEL 0-145 AND GO-145 AVIATION ENGINES



AN



LYCOMING, DIVISION - THE AVIATION CORPORATION

WILLIAMSPORT 38 PENNSYLVANIA, U. S. A.

Printed in U. S. A.

Price \$5.00

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DESCRIPTION

GENERAL The LYCOMING Model 0-145 and GO-145 engines are four cylinder, horizontally opposed, air-cooled aircraft engines. The A, B, and C series are identical in design with only such changes in construction as to permit the different horsepower ratings. The 0-145 is direct drive while the GO-145 has a reduction gearing the ratio of 27 to 17.

On the O-145 the A series is rated at 50-55 horsepower at 2300 R.P.M., B series at 65 horsepower at 2550 R.P.M. and C series at 75 horsepower at 3100 R.P.M. The GO-145 engines are of the same general design as the O-145 engines except those design features necessary to incorporate the reduction gear unit. The engine is rated at 75 horsepower at 3200 R.P.M. and a propeller R.P.M. of 2015.

The numeral following the suffixing letter denotes the equipment on engine, that is "1"single ignition, "2"-dual ignition, and "3"-dual ignition with generator and starter drive. All engines can be furnished or modified for the adoption of fuel pump if desired.

The O-145 crankshaft is forged chrome molybdenum steel and nitrided to produce a hard surface on journals and crankpins. Under no circumstance should any attempt be made to straighten this shaft.

The GO-145 crankshaft is forged chrome molybdenum steel, nitrided to produce a hard surface on the main and crankpin journals and is equipped with double, dynamic, dampeners to assure smooth operation of the engine at all speeds. Under no circumstance should any attempt be made to straighten this shaft.

This engine is licensed for manufacture, sale and for use under United States patents Nos. 2,103,643, and 2,112,984.

The crankshaft reduction gear and propeller shaft are forged nickel molybdenum steel with hardened, heavy spur type teeth to insure long life. The propeller flange is 6.00 inches in diameter, whereas, the propeller flange of the O-145 direct driven engines has a diameter of 5.50 inches.

The magneto timing on the various types is as follows:

O-145-A1, C1	28°	B.T.C.
0-145-B1	22°	B.T.C.
O-145-A2, A3, B2, B3	20°	B.T.C.
O-145-C2, C3	25°	B.T.C.
GO-145-C1	28°	
GO-145-C2, C3	25°	

The method of timing the engines is explained on page 23.

The text of the Instructions contained in Section III to XII inclusive are written around the O-145-A Series engines and when reference is made to A1, A2, or A3 series the same would apply to the respective B, C, or GO series engines, unless otherwise specified.

OIL RECOMMENDATIONS

	O-145-A Series	O-145-B and C Series	GO-145 Series
SUMMER (Temp. above 40°F.)	S.A.E. 30	S.A.E. 40	S.A.E. 40
WINTER (Temp. below 40°F.)		S.A.E. 30W	S.A.E. 30W

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SPECIFICATIONS

LYCOMING MODEL 0-145 ENGINES

Type-Four Cylinder, Direct Drive, Horizontally Opposed, Air-Cooled

SERIES	0-145-A1	0.145.A2	0.145.A3	0.146.B1	0.146.05	0.148.81	10 111 0	0.140.00	
Ingine Type Certificate	194	199	190	010	010	010	0.0	20.0010	5-1-1-0
tated Horse Power	50		201	10	219	112	012	017	017
Lind D D V	0000	00	00	00	99	89	16	16	16
101 M	2300	2200	2300	2550	2550	2550	3100	3100	3100
Contellare D D M / Economy	2100	2100	2100	2300	2300	2300	2700	2700	2700
Performance	2150	2150	2160	2450	2450	2450	2850	2860	2860
lore	3%	3%	3%	3%	3.5	35	3%	35	3%
Stroke	3%	315	3 14	316	3 14	314	3 14	116	1
Jompression Ratio	5.65:1	5.65:1	5.65:1	6.5:1	6.5:1	6.5:1	0.5:1	6.6:1	6.6.1
"Iston Displacement-Cubic Inch	144.5	144.5	144.5	144.5	144.5	144.5	144.5	144.6	144 5
load Temperature-Maximum 'F.	525*	525*	525*	525.	525.	625*	.988	1954	50K+
Jarrel Temperature-Maximum "F	825*	326*	325*	325.	325*	325°	825.	325*	325-
Dil Temperature-Maximum 'F.	220*	220*	220*	220*	220*	220*	220*	-020	-066
Auel Octane	22	73	73	22	73	28	22	e t	
7uel Consumption-Cruising					2	2	2	2	2
BHF/Hr. Maximum-Lbs.	-99	.55	22.	.50	60	.60	.60	.50	.60
011 Sump Capacity-Quarts	4-5	4-5	4-5	4-5	4-5	4-5	4-5	54	4-5
Dil Sump-Safe Quantity-Quarts	5 2	2%	2%	5%	24	5.4	214	3.12	24
Oil Pressure Minimum Idling-Lbs	15	16	16	15	16	15	15	16	16
Normal Operating-Lbs.	55-80	55-80	55-80	55-80	55-80	66.80	66.80	55.80	66.80
Crankshaft Rotation-Anti-Propoller End .	Clockwise	Clockwise	Clockwise	Clockwise	Clackwise	Clockwise	Clockwise	Clockwise	Clockwise
Valve Clearance-Intake and									
Doub Contraction of the contract	-910	.010	.015	.016.	.016	.015"	.015	.016"	.016"
DRIK Occurs-Degrees B.T.C.	-82	20.	20*	22.	20*	20*	28.	26°	26.
Tring Order	1-3-2-4	1-3-2-4	1-3-2-4	1-3-2-4	1-3-2-4	1-3-2-4	1-3-2-4	1-3-2-4	1-3-2-4
Spark Plugs-Champion .025" gnp	AY-4	AY-4	AY-4	474	AY-4	AY.4	AY-4	AY-4	AV-4
Engine Dry Weight including Carburetor, Magneto or Magnetos and Spark Plugs									
Lbs.	152	162	164	155	165.5	167.5	155	165.5	167.5
Propeller Hub and Baffles-Lbs.	5.33	5,33	5.33	5.33	6.33	5.33	5.33	5.33	6.33
Engine Total Weight-Lba.	157.33	167.33	169.33	160.33	170.83	172.83	160.33	170.83	172.83
Sudio Shielding-Added Weight-Lbs	-54	1.18	1.18	.64	1.18	1.18	.64	1.18	1.18
Carburetor-Marvel .	MA-2	MA.2							
Altitude Control Available	Available	Available	Available	Available	Available	Available	Available	Available	Avnilable
Magneto-Scintilla SF-4L	One	Two	Two	One	Two	Two	Ono	Two	Two
Pued Pump	Available								
Generator and Starter	None	None	Available	None	None	Available	None	None	Available
									transa a

* Model of carburetor same for all engines but jet size different.

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LYCOMING MODEL 0-145 AND GO-145 AVIATION ENGINES

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Figure 1

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LYCOMING MODEL 0-145 AND GO-145 AVIATION ENGINES

		1		N	lember Requir			red		
Part No.	PART NAME	AI	A2	A3	B1 C1	BZ	B3 C3	60 C1	100 C2	60 C3
45431	Gasket-Cylinder Head	8	8	8	8	8	8	8	8	8
45433	Seat-Valve Intake	4	4	4	4	4	4	4	4	4
45434	Stud	1997	24	3		33	12	18		100
	Magneto	1	4	4		4	4		4	4
	Generator or Cover		1	4			4			4
	Starter or Cover			0		0	0	0	0	0
45435	Stud_Rocker Box Cover	18	16	18	16	16	1.4	16	18	18
45436	Connection-Exhaust Pine	14	4	4	4	4	14	4	4	4
	(For alternate-See 45492 Connection)	1.	1		1.1	1	1.7			
45437	Connection-Intake Pipe	1			100		1	1		100
	Cylinder Head	4	4	4	4	4	4	4	4	4
	Oil Sump	4	4	4	4	4	4	4	4	4
45440	Seat-Valve Spring-Lower (Except C Series)	8	8	8	8	8	8			
40441	Seat-Valve Spring-Opper (Except C Series)	18	8	e e	8	8	8			
45448	Key_Valve (Darble Groove) (As Rea'd)	• •	0	0						
45444	Circlin-285 I.D. x .035 Dia	1						1.75		
	Valve Stems (For double Groove Valve only)	1				1				
	(As Required)	100	100		152	100	1.00	1.2	1.1	100
22025	Fuel Pump Plunger	1	1	1	1	1	1	1	1	1
45447	Bushing-Valve Rocker	- 8	8	8	8	8	8	8	8	8
45448	Shalt—Valve Rocker	4	4	4	4	4	4	4	4	4
40449	Value Rocker Shaft									
	Oil Sump (Refer to STD 453)	1 1	1	1	1	1	1	1	1	1
45450	Screw-Valve Adjusting	1 8	8	8	8	8	8	8	8	R
45451	Gasket-21/32 LD.x15/16 O.D.x.040, 030 thick	1 °		•		0	10	0	0	
	Valve-Rocker Shaft Plug	8	8	8	8	8	8	8	8	8
	Thermometer Hole Plug	i	i	i	i	i	II	11	i	i
45452	Stud-Cylinder Head	32	32	32	32	32	32	32	32	32
45453	Spring-Valve (B Engines Only)	-			8	8	8			
45455	Gasket-Cylinder Head-Inner	4	4	4	4	4	4	4	4	4
45456	XD Guide-Valve	8	8	8	8	8	8	8	8	8
45465	Seat-Exhaust Valve	4	4	4	4	4	4	4	4	4
- 45471	Seat-Valve Spring-Upper (Dual Springs)	1.3					0		0	
45470	(C Engine Unity)	í .			0	0	0	0	0	0
- 40412	(C Engine Only)							2		
- 45473	Spring_Valve_Outer (Dual Springs) (C Engs Only)	1			ŝ	18	i è	8	1 ê	ŝ
- 45474	Spring-Valve-Inner (Dual Springs) (C Engs. Only)	1			8	8	8	8	8	1 a
45475	Valve-Exhaust (Single Groove)	4	4	4	4	4	4	4	4	4
45476	Tube-Rocker Box Vent-Right	1 i	i	1 i	i	11	11	1	-	1.
	(External Ventilation System Only-Discontinued-					1	1			
	Ending with Engine No. 488)	1.3	1	1.1	1		Γ.		1	
45477	Tube-Rocker Box Vent-Left	1	1	1	1	1	1			
	(External Ventilation System Only-Discontinued-	1.2						1	1	
45 470	Ending with Engine No. 488)									
10110	(External Ventilation System Only, Discontinued	- 2	2	2	-	2	12			
	Ending with Engine No. 488)				1					
45479	Screw-Rocker Box Ventilation	4	4	4	4	4	4			
	(External Ventilation System Only-Discontinued-		-	1	10	1	1			
	Ending with Engine No. 488)									
45483	Gasket-Rocker Box Cover	4	4	4	4	4	4	4	4	4
	(Prior to Engine No. 489-Remove Center Section)	1	1		125	100		1.5		
45484	Plug-Generator Idler Shaft Hole	-		1			1			1
	(Not Required when using Generator Drive)		1	1	1.	i .				
45485	Cover—Rocker Box	- 4	4	4	4	4	1	4	1	4
45486	Gear-Grankshaft	-				1			1	
45400	Connection Exhaust Pine/(Sarary in Connection)	1	1	4	4	1.	1	1	4	1
40492	(Refer to Service Rullatin No. 102)									
45493	Key-Valve (Single Groove)	16	15	16	16	16	16	16	16	16
45501	Shaft Assembly-Oil Pump and Tachometer Drive	ĩ	10	1	Ĩi	1	1.0	٢.	1	1
0.000	(Discontinued-Ending with Engine No. 1373-	1			12			1		
	Superseded by part No. 45505)		1	1	100	100	1.00	1	1.5	
45502	Shaft Assembly-Oil Pump and Tachometer Drive	_ 1	1	1	1	1	1	11	1	1
	(Required with Fuel Pump Drive Only)			-						1
45505	Shaft Assembly-Oil Pump and Tachometer Drive	1	1	1	1	1	1	1	1	1
	(On single ignition engines prior to No. 1374									1
45514	use part No. 40001)	1.			4		1.	1.	1.	
45511	Cland Tashamatan Chaft	- 1	1	1	11	1	1	1	11	1
40012	Ouchd-Tacuolinetel, Digit		1	1	-	-	1	1	-	-
				1						
			18							15
	"-Parts not serviced as details	1	199							13
	-		1.5			1		1	1	
010000										

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Ce	ntrol surface m	ovements	Wing flaps	Up	0°	Dos	wn 16.5					
			Ailerons	Up	15.5°	Dov	vn 7º					
			Elevators	Up	25°	Dov	vn 13*	(stabilizer)	neutral at 0°)			
			Rudder	Right	23°	L	eft 18*	2000-001				
			Stabilizer (Leading E	dee) Up	1.5°	Dov	vn 3.5*	č.				
Se	rial Nos. eligib	le .	323 and up	-0-7-1	1440	200						
Re	quired equipm	ent	In addition to the per-	tinent rea	uired basi	ic equinmen	t specific	d in CAR	the following			
	21 100		items of equipment n	nust be in	stalled			- In Cruck	s are tonowing			
			2, 101, 201(c), 202(b)). 205(b).	206(b) 4	0100						
			(For night flying inst	mumont li	able or on	minglant to a	maidai	Iberninetion	of all alcounts			
			and instruments and i	flacher tur	and in the	pervarean to j	na liaht	monunación	t of all placates			
			required in addition t	o equipm	ent requir	ed by CAR	ng ngar 43).	or aurai (El	om) warning are			
SPECIF	ICATIONS PE	RTINENT TO	DALL MODELS									
Da	tum		25 19 in forward of a	vino leadi	ina edue e	wthosed of i	6III.et					
Le	veling means		Right hand engine he	oner tube	ALIRI -	ad M. 181 A	hinet.					
			is 00° to level (M. 19)	ALC: 1000	A M 190	nd M-ISLA	j; mont i	ace of from	spar under seat			
Ce	rtification basi		Tume Certificate No.	003 (CAD	n, m-180	- and M-180	, 33)					
Pre	duction basis		None Drive to orlolo	al and CAR	(3)			0.000000000	111021000			
Trougerour ouses			None. Phot to dergin	detailed inspection for workmonship, materials, and conformity with the process of								
			technical data and a check of the flight characteristics.									
Emileos		4-1-10										
installed.			or minus (-) sign preced	ing the wi	cight of a	n item indica	ates net o	change whe	n that item is			
installed.			mucates "Does no	c apply."								
		Approval	for the installation of all	items of e	quipment	listed herei	n has bee	en obtained	by the aircraft			
		manufactu	rer except those items pr	ecoded by	an asteri	sk (*). The :	asterisk	denotes that	t approval has			
		been obtai	ned by someone other the	an the aire	raft man	afacturer. A	n item n	tarked with	an asterisk may			
		not have b	een manufactured under	a FAA mo	onitored o	e approved o	quality o	outrol syste	m.			
Propeller	and Propeller A	consories							MINGLES			
1.	Deleted - April	28, 1950				SPICE.	ST-TOL	W-INLA	MIN. 22			
2.	Propeller-Sens	mich 66CB-52	or any other fixed		10.15	671.51		621.5				
	pitch wood pro	geller which is	cligible for the engine powe			(41.5)	200	(-21.5)				
	and speed and	which meets the	e following limits:									
	Static r.p.m. at	maximum perm	nissible throttle setting:									
	Not over	2085, not under	r 1890.									
	No additi	onal tolerance p	pennitted.									
	Diameter: Not	over 66 in., no	t under 64.5 in.									
3.	Propeller-Sens	enich 60LY44	or 60MY49 or		11 lb.	(-21.5)		(-21.5)	-			
	any other fixed	pitch wood pre	opeller which is eligible									
	for the engine ;	power and speed and which meets the										
	following limit	K.										
	Static r.p.m. at	p.m. at maximum permissible throttle setting:										
	Not over	2350, net under	r 2100.									
	No addits	enal tolerance j	permitted.									
	Dumeter: Not	over 60 in., no	t under 59 m.									
	(nem 401(c) re	quired)										
4.	ether front oite	arp es Lou ar e.	SLOG OF any		× 10.	(-21.5)		(-21.5)				
	other men pin	a wood propen	er witch is									
	which ments the	engine power a	ind speed and									
	Static ram at	e soliowing mil	inible through antipar									
	Not over	2220, not under	r 1995									
	No additi	enal tolerance a	emitted									
	Diameter: Not	over 63 in., not	tunder 62 in									
	(Item 401(d) re	quired).										
5.	Propeller-Flott	079 65A60 to 64	6, Sensenich		9 Ib.	272.2	623)	-	6210			
	65CK58 to 62.	or any other fix	ed pitch			69231	()	1000	(25)			
	wood propeller	which is eligib	le for the engine power									
	and speed and t	which meets the	following limits:									
	Static r.p.m. at	maximum perm	issible throttle setting:									
	Not over	2030, not under	1800.									
	No additi	onal tolerance p	semitted.									
	Diameter: Not	over 65 in., not	tunder 63.5 in.									